

References

¹ Ellen, C. H., "Influence of Structural Damping on Panel Flutter," *AIAA Journal*, Vol. 6, No. 11, Nov. 1968, pp. 2169-2174.

² Jordan, P. F., private communication.

Errata: "Analysis of the Optimum Two-Impulse Orbital Transfer under Arbitrary Terminal Conditions"

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[AIAA J. 6, 2145-2153 (1968)]

THROUGH my recent correspondence with R. A. Bass and E. A. McGinness of Bellcomm, Inc., Washington, D. C., an error was found in the coefficients C_3 and C_5 of the octic equation on p. 2147. The corrected coefficients should read:

$$C_3 = 4K^3(M_{02} - M_{01}) + 4K^2M_{01}M_{02}(N_{01} - N_{02}) + 2K^2(N_{01}P_{02} - N_{02}P_{01} + N_{01}M_{02}^2 - N_{02}M_{01}^2)$$

$$C_5 = 4K^2(N_{02} - N_{01}) + 4KN_{01}N_{02}(M_{01} - M_{02}) + 2K(M_{01}P_{02} - M_{02}P_{01} + M_{01}N_{02}^2 - M_{02}N_{01}^2)$$

Also a typographical error appears in Eq. (20), where the angle $\psi + \varphi_1$ should read $\psi + \varphi_i$.

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Erratum: "Axisymmetric Dynamic Snap-Through of Elastic Clamped Shallow Spherical Shells"

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RECENTLY more data about the critical load for dynamic snap-through of a clamped shallow spherical shell with $\lambda = 6$ under step loading have been obtained. It is found that Table 2 and Fig. 7 should be revised as follows:

Table 2 Snap-through loads p_c for various values of λ

λ	4.0	5.0	6.0	7.0	7.5	8.0	9.0	10.0	11.0	13.0
p_c	0.45	0.49	0.61	0.56	0.50	0.44	0.39	0.42	0.50	0.42

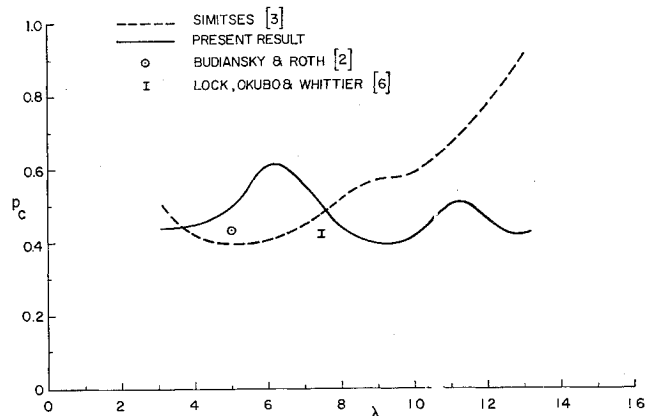


Fig. 1 Comparison of present values of P_c with the previous results.

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